17th WRMISS Conference Austin, USA

September 4-6, 2012

ISSCREM: International Space Station Cosmic Radiation Exposure Model



S. El-Jaby, B. Lewis Royal Military College of Canada



L. Tomi Canadian Space Agency



L. Sihver Chalmers University of Technology



T. Sato Japan Atomic Energy Agency



K. Lee, S. Johnson Space Radiation Analysis Group, NASA

Outline_

- Motivation
- Model Development
- Model Benchmarking
- Effective Dose Simulations
- Summary



Motivation.

Radiation exposure assessment is part of mission planning aboard the ISS (Necessitated by the Radiation Health Working Group (RHWG))



ISS Low-Earth Orbit (LEO)

- 90 minute orbit
- 340 to 420 km altitude
- Inclination of 51.6°

Main Sources of Radiation in LEO

- Trapped Radiation (TR)
- Galactic Cosmic Radiation (GCR)

Motivation.

To develop a model to predict the radiation dose space-crew can expect to get for a given mission aboard the ISS.

Specifically:

- Give measure of the dose equivalent and the effective dose space-crew get from GCR and TR exposure.
 - Dose equivalent is an operational quantity (radiation weighted)
 - Effective dose is a protection quantity (radiation and tissue weighted)
- Base model on operational data collected aboard the ISS (empirical model)
- Correlate the model to physical phenomena occurring in LEO.

Operational Data

• NASA has been operating a Tissue Equivalent Proportional Counter (TEPC).



- TEPC data available from 2000 to 2010.
- 2001 and 2008 data used in model development.
- Rest of data used in model benchmarking.

Data contains:

- Dose equivalent rate (µSv min⁻¹)
- ISS position
- Radiation Flag (GCR or trapped radiation)
- TEPC detector location

GCR parametric model

- GCR dose is anti-coincident with solar-cycle and dependent on the ability of GCR ions to penetrate magnetosphere.
- The measured TEPC dose equivalent rate from GCR exposure was correlated to the cutoff rigidity parameter interpolated from ISS state vectors.
- Cutoff rigidity maps were obtained from Smart and Shea RCINUT3 code.



7



Trapped radiation parametric model

- TR dose is anti-coincident with solar activity due to atmospheric density effects inside the SAA.
- NRLMSIS-00 atmospheric density model was used.
- Solar activity was accounted for by solar radio flux at 10.7 cm (F10.7) and Ap magnetic index.
- Model was developed for the TEPC detector located at SM-327 with the detector orientated perpendicular to the velocity vector.





9

South Atlantic Anomaly delineation

Region of South Atlantic Anomaly for 2008 (0.6 μ Sv min⁻¹ Dose Equivalent Rate Threshold)





11

Graphical User Interface

 International Space Station Cosmic Radiation Exposure Model (ISSCREM)



Benchmarking

% Difference in Cumulative Dose for GCR and TR Dose Predictions (TEPC Located at SM-327)



14 Benchmarking

GCR Dose Predictions (2009)



Date



15 Benchmarking



from February 13 to March 30, 2009 TEPC detector orientation at SM-327 from 17 June, 2009 to 8 July, 2009 TEPC detector orientation at SM-327 from July 9 to August 21, 2009

Effective Dose.

• Want to get measure of the effective dose which is a protection quantity (biologically relevant dose).

Protection
Quantity
$$\leftarrow E = \frac{E}{H^*(10)} H_{TEPC} \longrightarrow \begin{array}{c} \text{Operational} \\ \text{Quantity} \end{array}$$



Effective Dose_

How to calculate E/H^{*}(10)?

• Particle and Heavy Ion Transport Code System (PHITS) Monte Carlo code used to transport radiation through a simple representation of the ISS.



Effective Dose.

How to calculate E/H^{*}(10)?

- Primary and secondary particle flux distribution inside ISS modelled.
- Example flux distribution inside ISS (GCR at solar minimum incident on cylinder with 4 cm wall thickness).



Effective Dose.

What was determined?

- Simulated trapped radiation H*(10) and E within factor of 2 of measured data and published works.
- Neglecting heavy-ion contribution from GCR exposures results in underestimation of H*(10) and E as confirmed before¹.
- Thicker shielding increases GCR H*(10) and E but decreases trapped radiation H*(10) and E.
- TEPC dose equivalent can be used as conservative estimate of effective dose based on simulations.

Trapped RadiationGCR $\frac{E}{H^*(10)} = 0.25$ to 0.3 $\frac{E}{H^*(10)} = 0.35$ to 0.65

• Taking the ratio of the effective dose predicted by Sato² to the measured TEPC dose equivalent results in E/H(TEPC)= 0.5 to 0.6.

[1] Z. Kolísková (Mrázová), L. Sihver, I. Ambrožová, T. Sato, F. Spurný, and V. A. Shurshakov, Adv. Space Res. 49, 230-236 (2011 [2] T. Sato, A. Endo, L. Sihver, K. Niita, Radiat. and Environ. Biophysics, Vol. 50, pp. 115 – 123, 2011

Effective Dose_

Comparisons made against a more complex geometry.

- Modeled complex geometry at 2 cm and 10 cm aluminum wall thicknesses.
- Simulated E/H*(10) for trapped radiation and GCR at solar minimum conditions.
 Poisk Docking Module



Effective Dose.

Trapped radiation induced flux distribution at solar minimum incident on simple cylinder and complex representation of Service Module at 5.4 g cm⁻² aluminum.



Effective Dose.

GCR flux induced distribution at solar minimum incident on simple cylinder and complex representation of Service Module at 5.4 g cm⁻² aluminum.



Effective Dose_

Comparisons made against a more complex geometry.

		Trapped Proton			Galactic Cosmic Radiation		
Wall Thickness (g cm ⁻²)	Geometry	H*(10) (Sv d⁻¹)	E (Sv d ⁻¹)	E/H*(10)	H*(10) (Sv d⁻¹)	E (Sv d ⁻¹)	E/H*(10)
5.40	Complex	8.6e-04	2.2e-04	0.25	2.0e-04	1.3e-04	0.63
	Simple	8.5e-04	2.2e-04	0.25	1.0e-04	6.3e-05	0.63
27.0	Complex	1.5e-04	4.3e-05	0.29	4.3e-04	1.5e-04	0.36
	Simple	1.4e-04	4.1e-05	0.29	2.1e-04	7.4e-05	0.36

- E/H^{*}(10) for GCR and TR remained same.
- Trapped radiation doses (ambient and effective) remained the same with complex geometry but GCR doses (ambient and effective) increased.
- Demonstrates that local shielding effects and module geometry play a role in modeling the amount of dose received (most work is done with simple geometries!)

Neutron and Proton Response Function



*Bubble Technology Industries, Report on Characterization of the Space Bubble Detector Spectrometer, March 31st 2010 (Detector sensitivity of 0.1 bubble μ Sv⁻¹)

*Takada et al, *Measured Proton Sensitivities of Bubble Detectors*, Radiation Protection Dosimetry, Vol. 111 (2), 181-189)

PHITS Analysis: Application to BD Interpretation



GCR induced flux distribution at solar minimum incident on simple cylinder representation of Service Module at 10.8 g cm⁻² aluminum.

PHITS Analysis: Application to BD Interpretation



GCR and TR flux induced distribution at solar minimum inside simple cylinder geometry of Service Module at 2, 4, 6, and 10 cm wall thicknesses.

28

PHITS Analysis: Application to BD Interpretation



29

PRELIMINARY RESULTS



GCR flux induced distribution at solar minimum inside SM-327 CQ (complex geometry) and inside the air-cavity of the Matroshka phantom.

Conclusions.

Principle Code

- Developed a predictive model which relates ISS crew dose rate to physical phenomena in LEO.
- Can predict GCR and trapped radiation TEPC dose equivalents to within $\pm 10\%$ and $\pm 20\%$ on a daily basis and for a total mission.
- Accuracy is robust over a solar cycle and shown to behave well given variations in solar activity input parameters (based on a sensitivity analysis).
- Model has been implemented into a user-friendly software.

Protection Quantity

- PHITS Monte Carlo simulations have demonstrated that the TEPC dose equivalent is an excellent operational quantity.
- Complex geometry and shielding shown to be significant modifier of effective doses but not E/H^{*}(10) ratio.

Conclusions_

Bubble Detector Response

- Over the operating energy-response range of the BD, neutrons are the principle component observed in the device (as supported by PHITS calculations)
- Preliminary results for GCR induced flux distribution inside Matrohska air cavity at solar minimum also suggest neutrons are still the dominant component.
 - This point is also detailed in proton and heavy ion calculations performed in: B. J. Lewis, M. B. Smith, H. Ing, H.R. Andrews, R. Machrafi, L. Tomi, T. J. Matthews, L. Veloce, V. Shurshakov, I. Tchernykh and N. Khoshooniy, "Review of Bubble Detector Response Characteristics and Results from Space", Radiation Protection Dosimetry, doi: 10.1093/rpd/ncr358, September 1, 2011.
- On-going work includes:
 - Model experimental setup of Matroshka setup (BD placement) and compare to measurements.
 - Model target fragmentation in Matroshka phantom and BD

Future Work_

- Implement ISSCREM into lower-level computer language.
- Implement SPEs into model.
- As new TEPC is collecting data, scale the current and new TEPC to determine localized shielding conditions.
- Improve trapped model as more data is gathered.
- Model response of TEPC to mixed radiation field (heavy-ions).
- Improve model of the ISS shielding distribution for use into Monte Carlo simulations.



Supplemental Slides



Motivation.

State-of-the-Art Models

- A qualitative approach is taken to predict the expected dose space-crew are to receive aboard ISS.
- Current efforts are focused on modelling the radiation environment outside and inside the ISS.
 - HZETRN
 - SHIELDOSE
 - EVADOSE
 - DESIRE

- CREME (GCR Environment)
- AP8/AE8 (Trapped Environment)

- Empirical models have been developed for predicting dose aboard U.S. Space Shuttle and Mir Space Station.
- No truly predictive model exists for ISS mission planning.



Why use 2001 and 2008 TEPC data for model development?



- Sun follows 11 year cycle of solar activity.
- 2001 represents solar maximum conditions while 2008 represents solar minimum conditions.



Why is solar activity important for GCR dose?



- Solar wind extends beyond the solar system and acts as barrier for GCR ions.
- The Earth's magnetosphere deflects incoming ions.

Why is solar activity important for trapped radiation dose?



- South Atlantic Anomaly (SAA) is a dip in the Earth's magnetic field.
- Trapped radiation exposure is limited to within the SAA.
- Increase in solar activity results in increased atmospheric density and less trapped radiation dose.


Model Development

What other factors do we need to consider?



- ISS modules have different shielding distributions due to different construction and distribution of equipment inside them.
- Adjacent modules can also influence shielding.



Benchmarking.

Sensitivity Analysis

- Varied solar activity input parameters to test the effect on predicted dose.
 - Varied F10.7 by up to ±50%.
 - Varied Ap magnetic index from quiet to very disturbed conditions.
 - Varied U by up to ±20% as well as extreme conditions (i.e. solar max. or min.).
 - Varied the extent of SAA.
- Space Weather Prediction Center offers 45 day lead-time predictions of F10.7 and Ap magnetic index.
- ISS orbit is well defined.



Benchmarking_____

		% Difference from Measured Dose			
Dose Type	% Variation from True F107	2000	2007	2009	2010
	-50	-43	-34	0	-10
	-20	-5	-14	2	4
Total TD	-10	5	-6	4	12
	+10	10	5	13	22
	+20	10	9	18	26
	+50	10	20	30	36
	-50	-30	-29	-1	-13
	-20	-10	-14	2	-7
According Dass TD	-10	0	-7	4	0
Ascending Pass TK	+10	0	0	11	7
	+20	0	4	14	13
	+50	0	7	22	20
	-50	-70	-41	-9	-21
	-20	-30	-22	-3	8
Descending Dass TD	-10	-20	-14	2	13
Descending Pass TR	+10	0	-3	12	23
	+20	0	3	16	26
	+50	0	14	27	36



Benchmarking____

		% Difference from			
		Measured Dose			
Dose Type	Ap Index	2000	2007	2009	2010
	400	14	27	43	41
	100	14	16	25	29
Total TR	50	14	11	20	25
	25	14	6	16	22
	0	5	-6	6	12
	400	0	7	31	23
	100	0	4	19	17
Ascending Pass TR	50	0	0	15	10
_	25	0	0	12	10
	0	0	-7	6	0
	400	0	30	41	38
	100	0	11	23	28
Descending Pass TR	50	0	5	18	26
	25	-10	0	14	23
	0	-10	-14	3	15



Benchmarking____

	% Difference from Measured Dose				
% Variation from True	2000	2002	2007	2009	2010
-20	_9	-5	N/A		N/A
-10	-7	-4	3		2
+10	0	0	0	N/A	0
+20	N/A	N/A	4		3
Previous Year	N/A	N/A	4		2
Solar Maximum	-7	-3	11	11	11
Solar Minimum	-17	-13	3	3	2



Benchmarking____

		% Difference from Measured Dose			
Dose Type	SAA Contour Line (µSv min ⁻¹)	2000 2007 2009 207			
	0.6	-4	-4	3	-3
GCR	0.3	0	1	7	2
	0	2	3	9	4
	0.6	10	0	9	17
Total TR	0.3	10	-2	7	16
	0	10	0	8	14
	0.6	0	-4	7	3
Ascending Pass TR	0.3	0	-7	4	3
	0	0	-4	4	3
	0.6	0	-8	6	18
Descending Pass TR	0.3	0	-8	6	18
	0	0	-11	6	18



Effective Dose.

Input Trapped Radiation Flux Distribution

• AP8 model used to simulate trapped protons at solar maximum and minimum conditions.





Effective Dose.

Input GCR Flux Distribution

• CREME96 model used to simulate GCR at solar maximum and minimum conditions.





Effective Dose_

Trapped radiation effective-to-ambient dose equivalent conversion factors at solar maximum and solar minimum conditions incident on a simple aluminum cylinder as a function of wall thickness.

	Solar Maximum			Solar Minimum		
Wall Thickness (g cm ⁻²)	H*(10) (Sv d ⁻¹)	E (Sv d ⁻¹)	E/H*(10)	H*(10) (Sv d ⁻¹)	E (Sv d ⁻¹)	E/H*(10)
5.40	4.23e-04	1.26e-04	0.30	8.52e-04	2.17e-04	0.25
10.8	2.63e-04	8.21e-05	0.31	4.69e-04	1.31e-04	0.28
16.2	1.76e-04	5.65e-05	0.32	2.92e-04	8.52e-05	0.29
27.0	9.40e-05	2.93e-05	0.31	1.43e-04	4.13e-05	0.29



Effective Dose_

GCR effective-to-ambient dose equivalent conversion factors at solar maximum and solar minimum conditions incident on a simple aluminum cylinder as a function of wall thickness.

	Solar Maximum			Solar Minimum		
Wall Thickness (g cm ⁻²)	H*(10) (Sv d ⁻¹)	E (Sv d ⁻¹)	E/H*(10)	H*(10) (Sv d ⁻¹)	E (Sv d ⁻¹)	E/H*(10)
5.40	5.98e-5	3.92e-5	0.65	1.00e-4	6.34e-5	0.63
10.8	7.94e-5	4.22e-5	0.53	1.28e-4	6.71e-5	0.52
16.2	9.90e-5	4.47e-5	0.45	1.56e-4	7.01e-5	0.45
27.0	1.37e-4	4.84e-5	0.35	2.07e-4	7.42e-5	0.36



Effective Dose.



Effective Dose.



Pirs Docking Module

Complex representation of Zvezda Service Module (SM)⁽⁶⁵⁾ including Crew Quarters (SM-CQ) and Working-Quarters (SM-WQ) and neighboring Zarya Control Module (CM) ⁽⁶⁶⁾, and Poisk and Pirs Docking Modules (PDM) ⁽⁶⁷⁾. Dimensions of modules in meters (length I, diameter d): SM-CQ (5.07,4.34), SM-WQ (3.57,3.22), CN (2.23,2.02), CM(7.98,4.1), PMA(1.91,2.4), PDM(3.96,2.54). Length of connecting cones in meters: C1(1.15), C2(1.09), C3(1.74), C4(0.87), C5(0.95).











2007 Original.



2007 Adjusted

































2011 SM-327 Adjusted




2011 SM-327 Adjusted.

