

# A Directional Trapped Proton Model for the International Space Station Orbit



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**WRMISS20 Workshop**  
**Cologne Germany, September 08 - 10, 2015**

# Outline

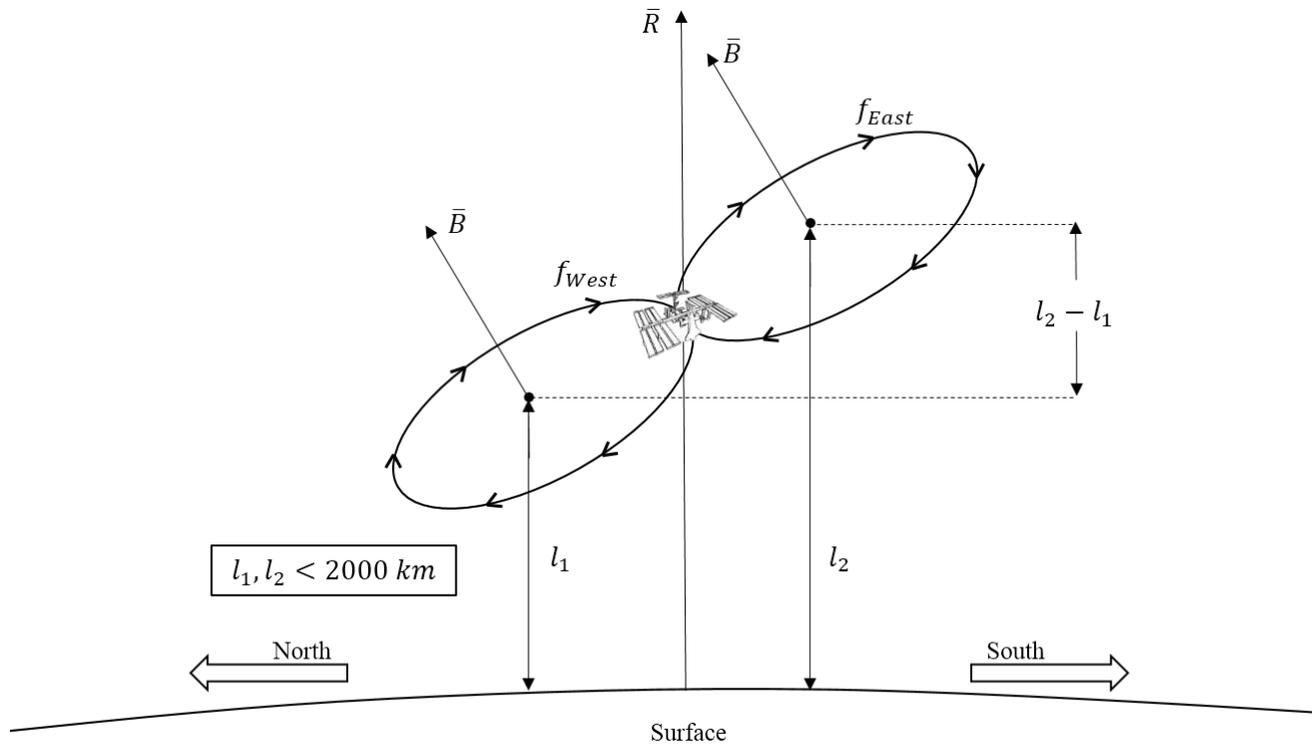


- Describe an angular model to account for proton flux anisotropy (directionality) at Low Earth Orbit (LEO) inside/outside the South Atlantic Anomaly (SAA) region
- Using US Air Force Tri-Service Experiment 5 (AF-TSX5) satellite measurements and the omni-directional trapped proton models AP8/AP9, validate the angular model at three locations within SAA
- Using Crew Quarter (CQ) of Zvezda Service Module (SM) ray-traced geometry of ISS, and the trapped models AP8/AP9, apply the angular model to proton anisotropy **dosimetric simulation** of ISS within SAA, and compare the simulation with ISS daily measurement
- Summary

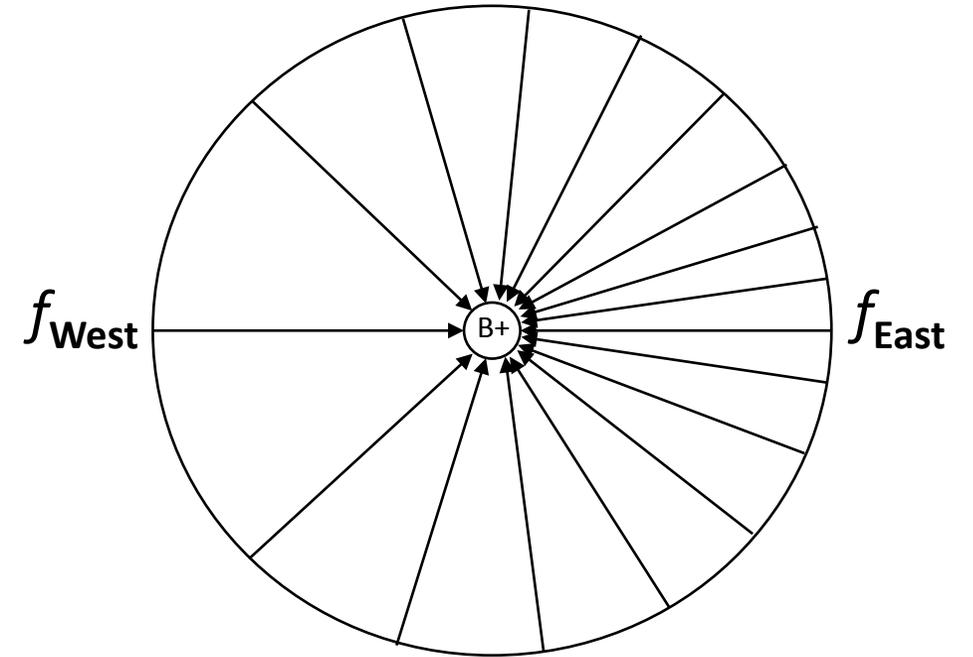
# “East – West effect” of Proton Anisotropy within SAA (100 – 2000 km)



$$\text{flux}_{\text{East}} > \text{flux}_{\text{West}}$$

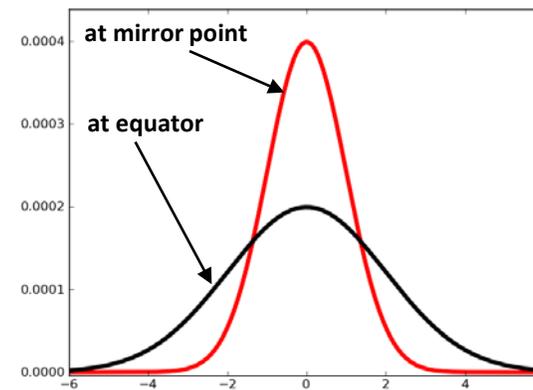
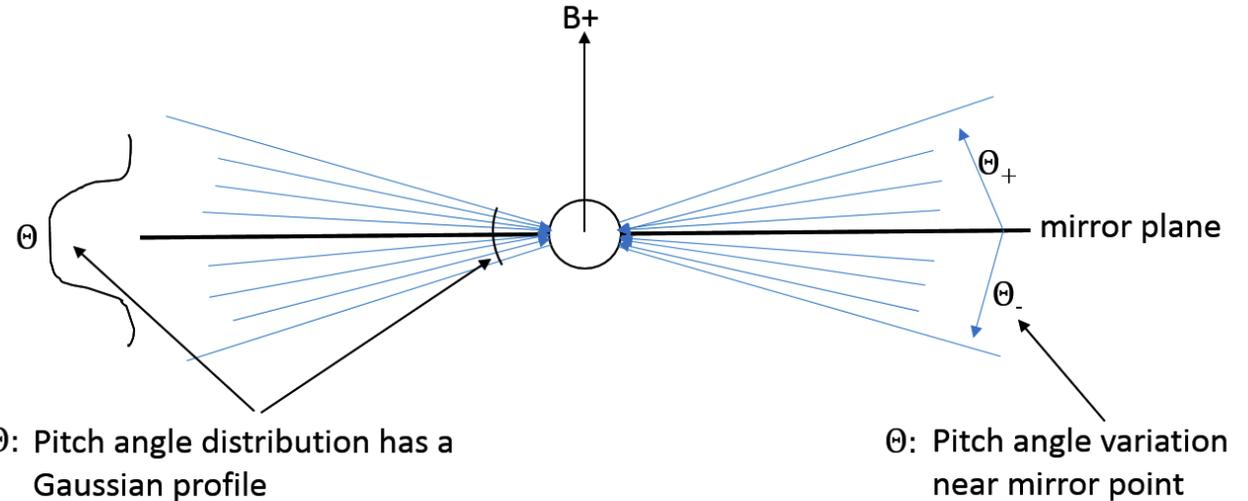
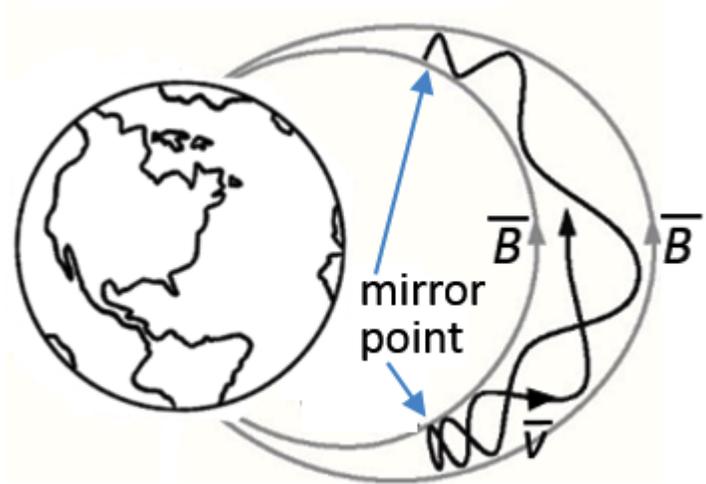


Side view



Top view

# “Pitch Angle Distribution Effect” of Proton Anisotropy Near Mirror Point within SAA



# Combined “East - West” and “Pitch Angle” Distribution Effects Formalism Describing Proton Anisotropy within SAA



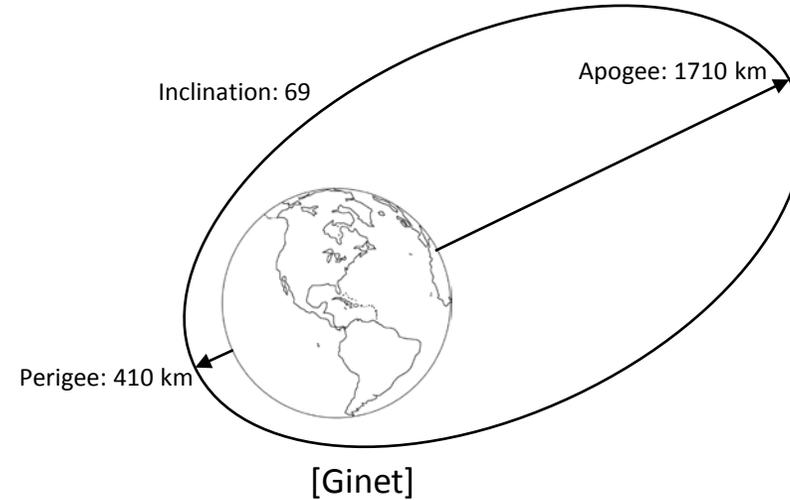
$$\begin{array}{c}
 \text{3D} \quad \text{1D} \quad \text{2D} \\
 \underbrace{\hspace{1.5cm}} \quad \underbrace{\hspace{1.5cm}} \quad \underbrace{\hspace{4.5cm}} \\
 f(E, \theta, \phi) = \underbrace{[f(E)]}_{\text{1D}} * \left[ \frac{\exp\left(-\frac{\left[\frac{\pi}{2} - \theta\right]^2}{2\sigma^2}\right)}{\sin\theta \sqrt{2\pi}\sigma \operatorname{erf}\left(\frac{\pi}{2\sqrt{2}\sigma}\right)} \right] * \left[ \frac{\exp\left(\frac{r_g \cos I \sin \theta \sin \phi}{h_0}\right)}{2\pi \sum_{n=0}^{\infty} \frac{1}{n!^2} \left(\frac{r_g \cos I \sin \theta}{2h_0}\right)^{2n}} \right] \\
 \uparrow \quad \quad \quad \uparrow \quad \quad \quad \uparrow \quad \quad \quad \uparrow \\
 \text{Directional} = \text{Omni} * \text{Pitch angle distribution} * \text{East-West distribution}
 \end{array}$$

- $h_0$  (300 – 2000 km) is atmospheric scale height (ISS = 80 km)
- $I$  is magnetic dip angle between B field and Earth surface (ISS  $I = 42^\circ$ )
- $\sigma$  is pitch angle standard deviation ( $h_0, I, R, \text{ISS } \sigma = 8^\circ$ )
- $r_g$  is proton gyro-radius (km)
- $\phi$  is azimuth (0 -  $2\pi$ )
- $\theta$  is pitch angle near mirror point

# AF-TSX5 Satellite Orbit Description

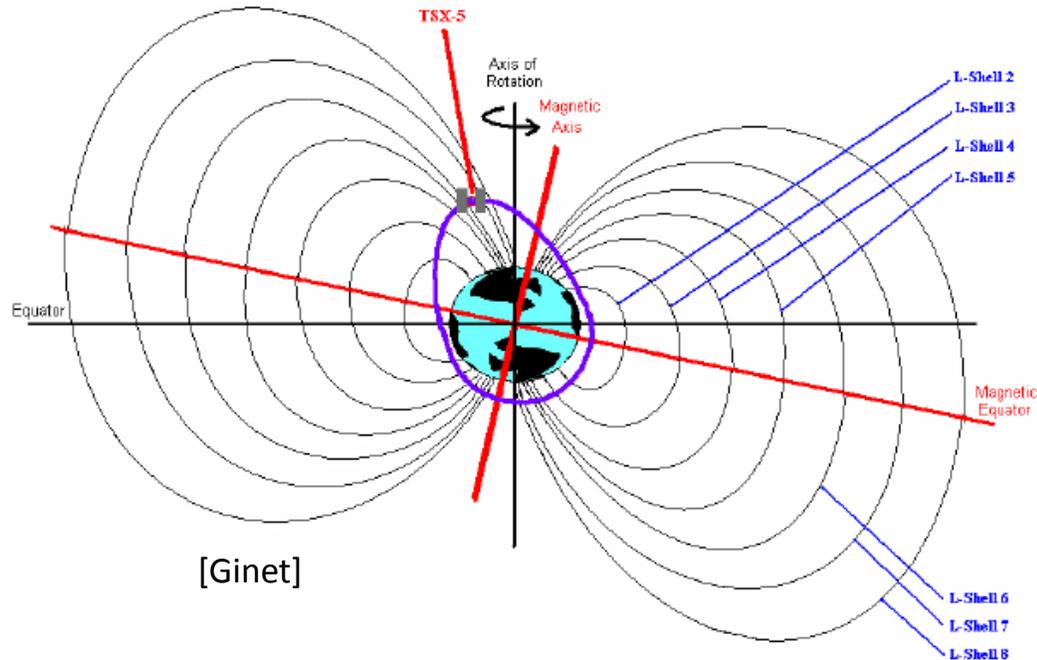


[Ginet] AF-TSX5 satellite (2001 - 2006)



- $69^\circ$  inclination allows coverage of many L shells (2 - 8) in a single orbit
- 410 - 1710 km altitude range allows coverage of atmospheric density variation within thermosphere and low exosphere (ionosphere)

# AF-TSX5 Compact Environment Anomaly Sensor (CEASE) Particle Detector

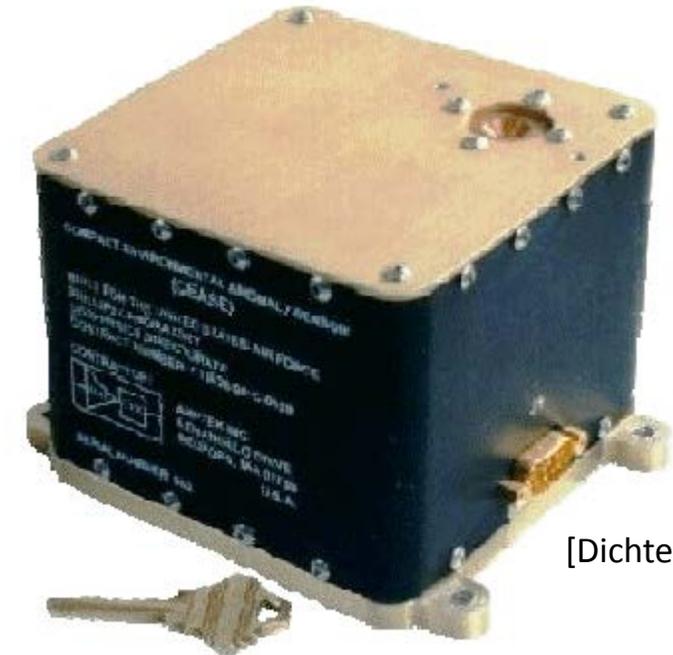


[Ginet]

Single orbit crossing of AF-TSX5 in B - L geomagnetic coordinate for L shells 1 - 8

Ginet, et al., Proton flux anisotropy in low earth orbit, IEEE Trans. Nucl. Sci. 54 (6) (2007) 1975-1980

Dichter, et al., Compact environmental anomaly sensor (cease): a novel spacecraft instrument for in situ measurement of environmental conditions, IEEE Trans. Nucl. Sci. 45 (6) (1998) 2758-2764.



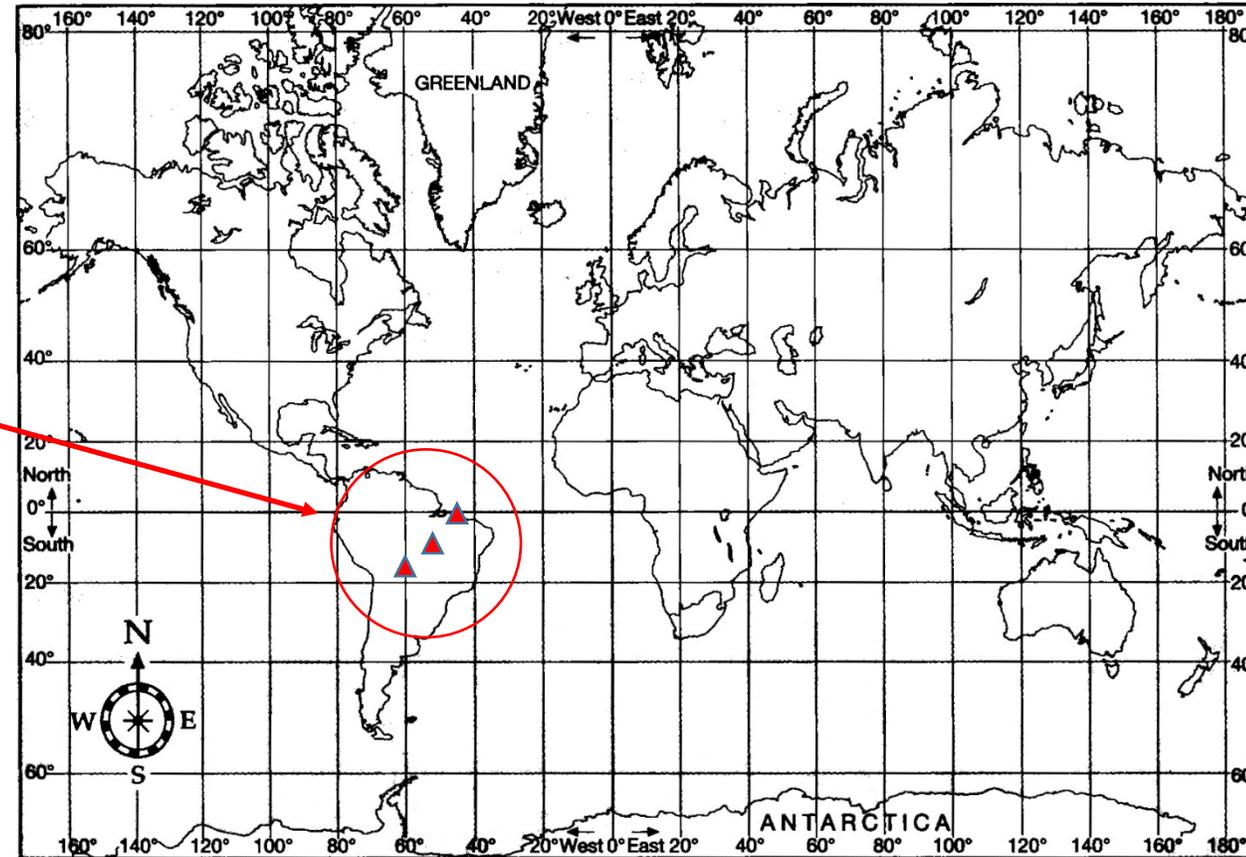
[Dichter]

CEASE (Compact Environment Anomaly Sensor) instrument. Small, light weight and low power particle detector (1.7 W)

# AF-TSX5 Compact Environment Anomaly SEnsor (CEASE) Particle Detector Data Acquisition Attitude



Here, longitude and latitude are fixed and altitude changes between 400 - 1700 km



SAA orbital crossing of AF-TSX5 for L shells 1 - 8

# Proton Anisotropy within SAA at a Geographic Point with a Specific Look Direction



$$f(E, \theta, \phi) = [f(E)] * \left[ \frac{\exp\left(-\frac{\left[\frac{\pi}{2} - \theta\right]^2}{2\sigma^2}\right)}{\sin\theta \sqrt{2\pi\sigma} \operatorname{erf}\left(\frac{\pi}{2\sqrt{2}\sigma}\right)} \right] * \left[ \frac{\exp\left(\frac{r_g \cos I \sin\theta \sin\phi}{h_0}\right)}{2\pi \sum_{n=0}^{\infty} \frac{1}{n!^2} \left(\frac{r_g \cos I \sin\theta}{2h_0}\right)^{2n}} \right]$$

Becomes constant
Has only one variable

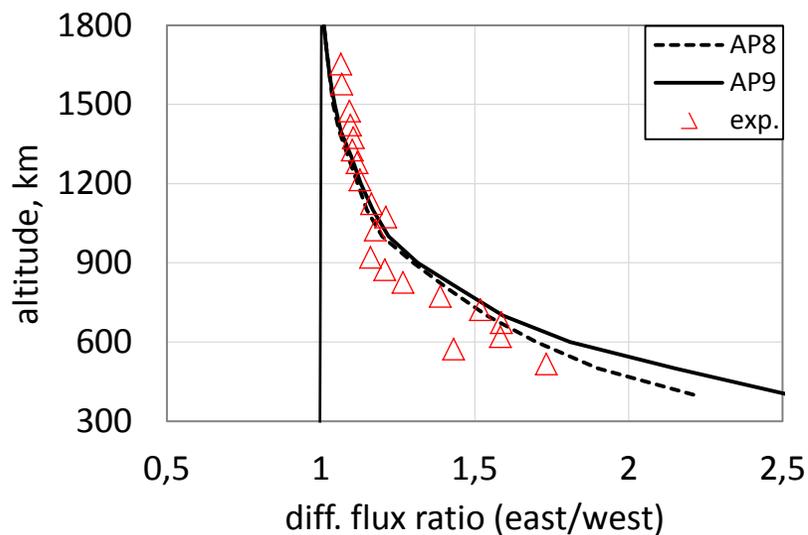
= are constants

= are variables

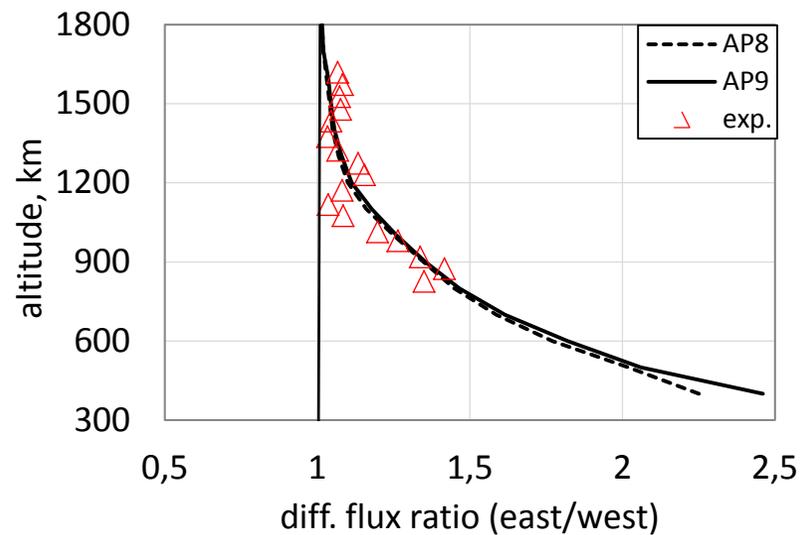
$$r_{g,p} \propto \frac{E_p}{B}$$

I want to look at only E and W directions, so I make azimuth angle  $\phi$  to be either  $90^\circ$  or  $270^\circ$  degree

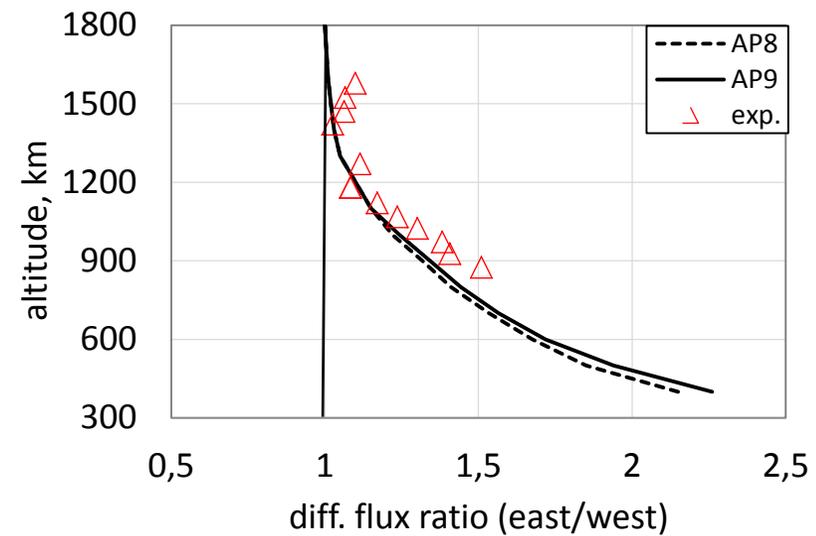
# AF-TSX5 CEASE Differential Flux (40 MeV) Proton Validation of AP8/AP9 Trapped Models



40 MeV flux ratios (east/west) at 18°S, 300° E

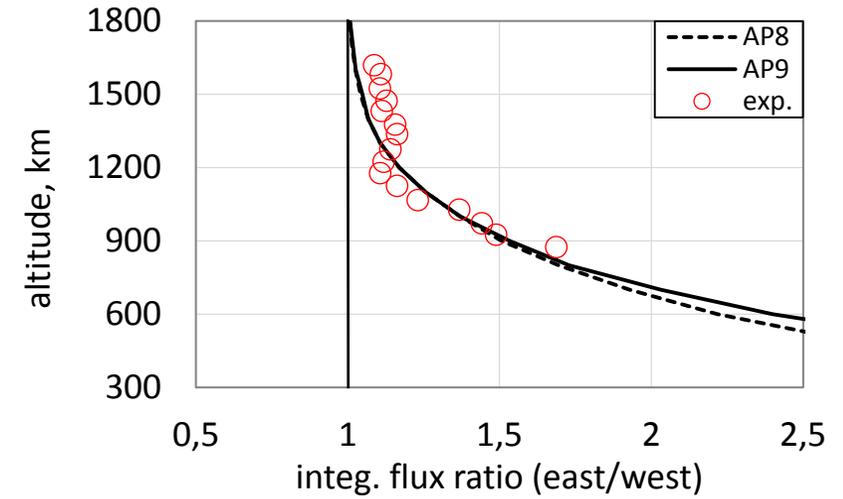
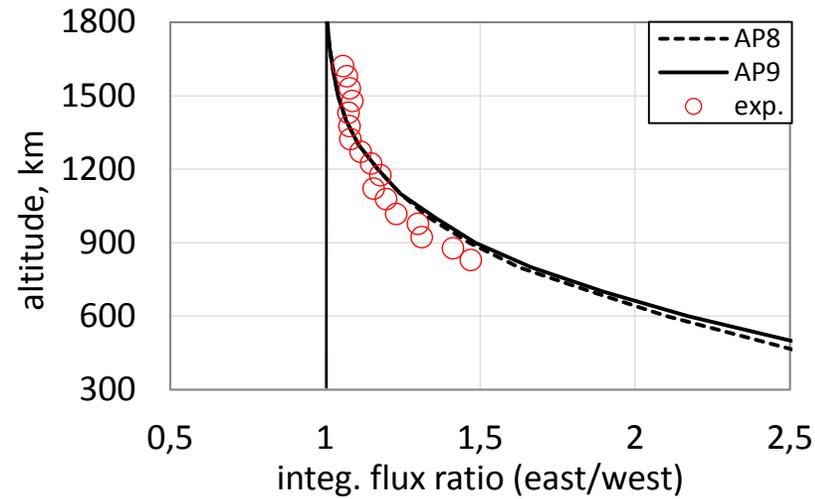
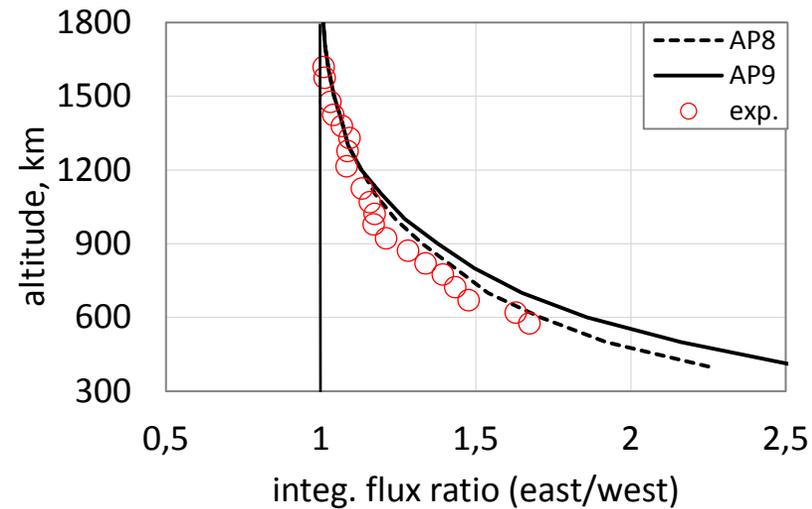


40 MeV flux ratios (east/west) at 9°S, 309° E



40 MeV flux ratios (east/west) at 0°S, 315° E

# AF-TSX5 CEASE Integral Proton Flux (>40 MeV) Validation of AP8/AP9 Trapped Models

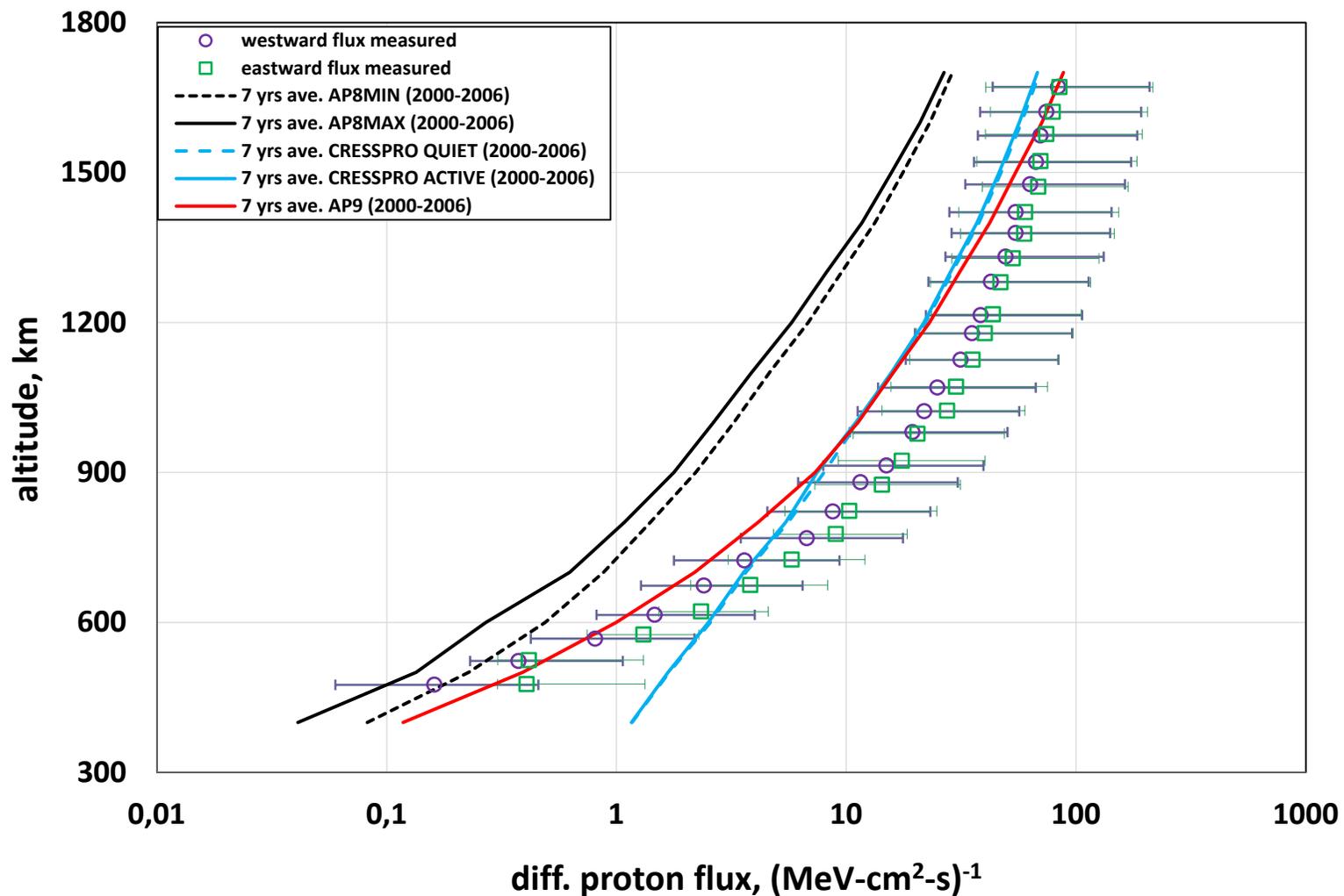


>40 MeV flux ratios (east/west) at 18°S, 300° E

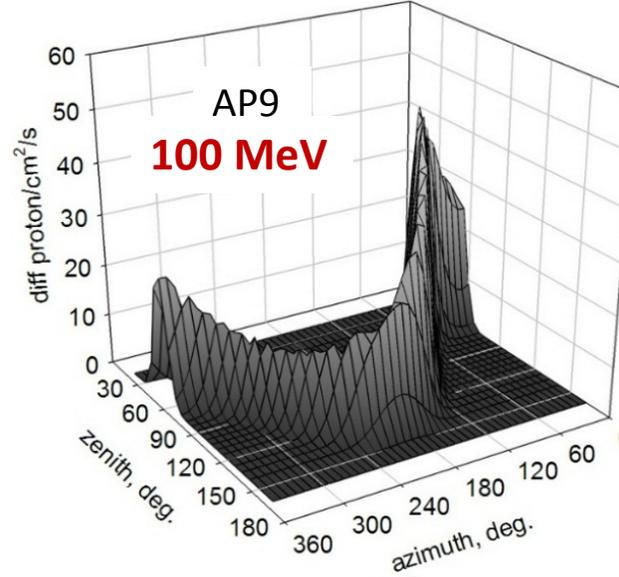
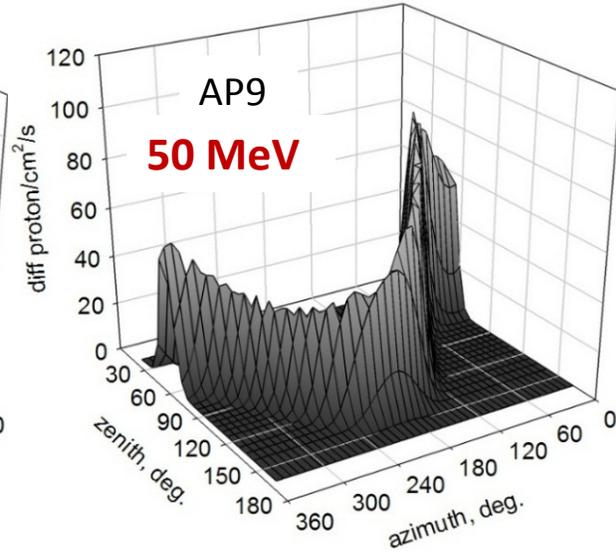
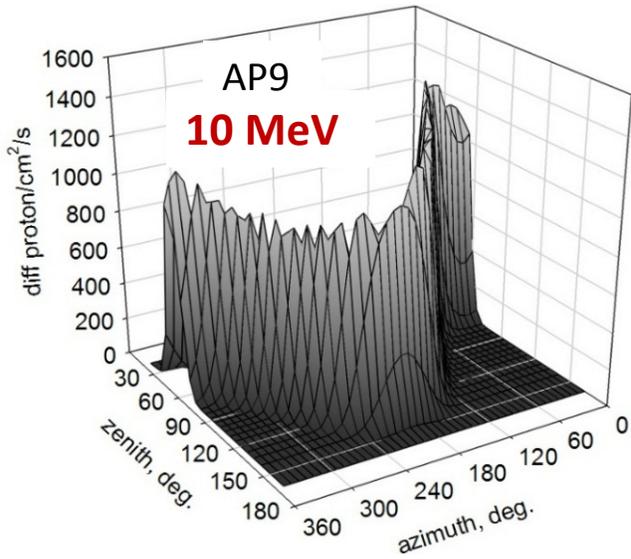
>40 MeV flux ratios (east/west) at 9°S, 309° E

>40 MeV flux ratios (east/west) at 0°S, 315° E

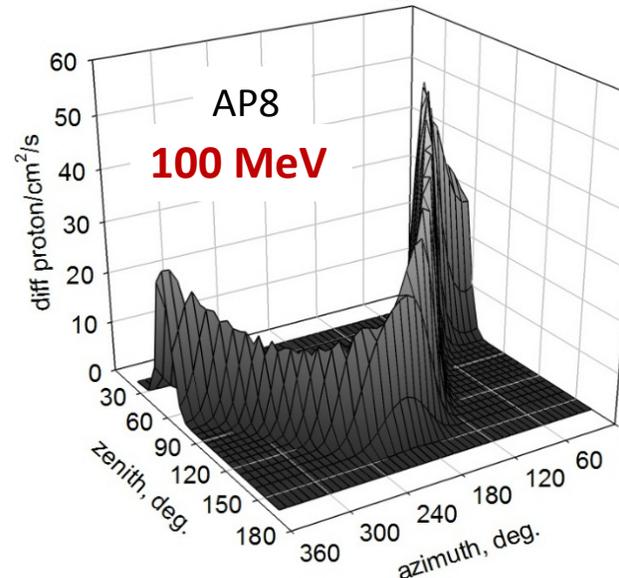
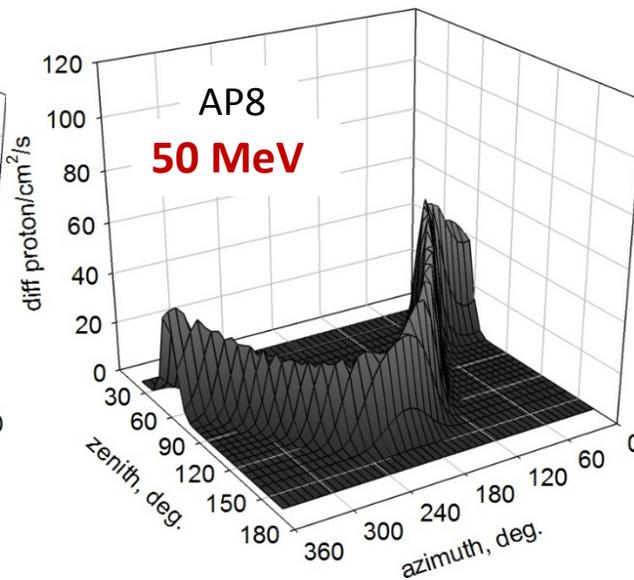
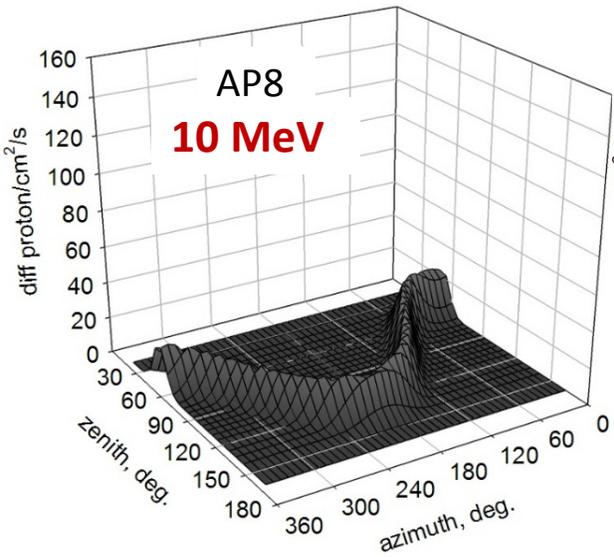
# AF-TSX5 CEASE Validation of AP8/AP9/CRESSPro Trapped Models



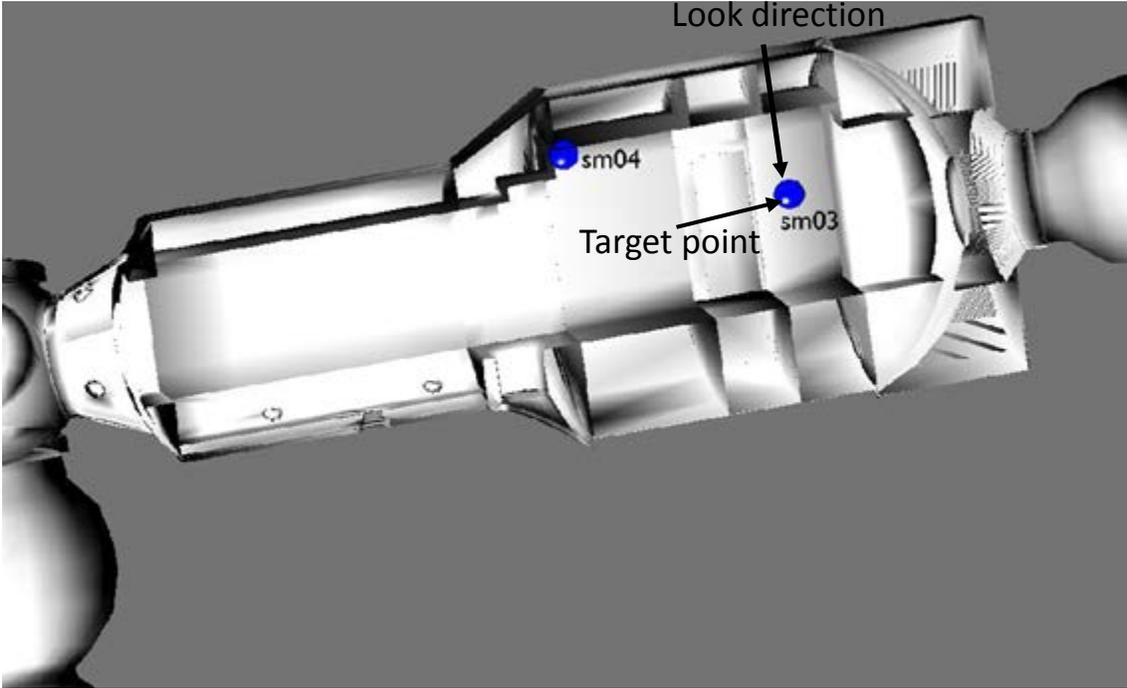
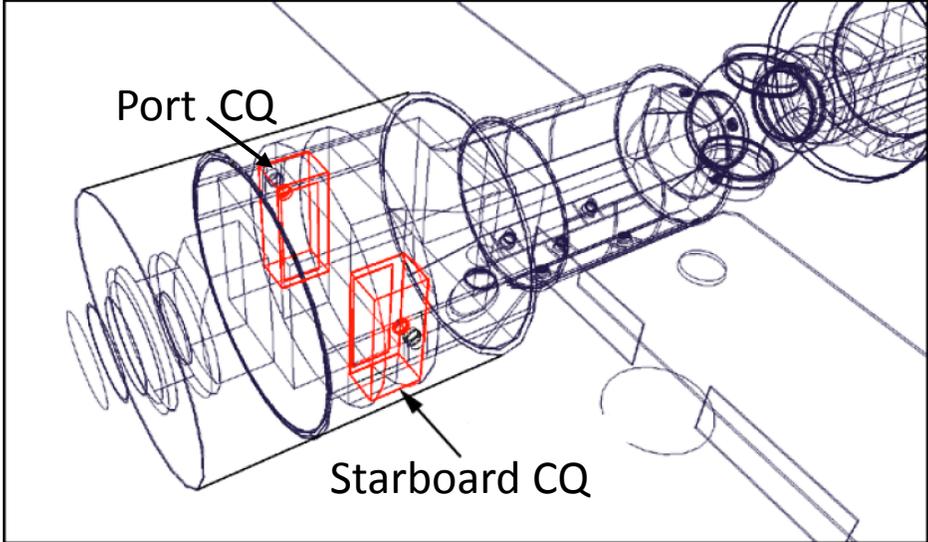
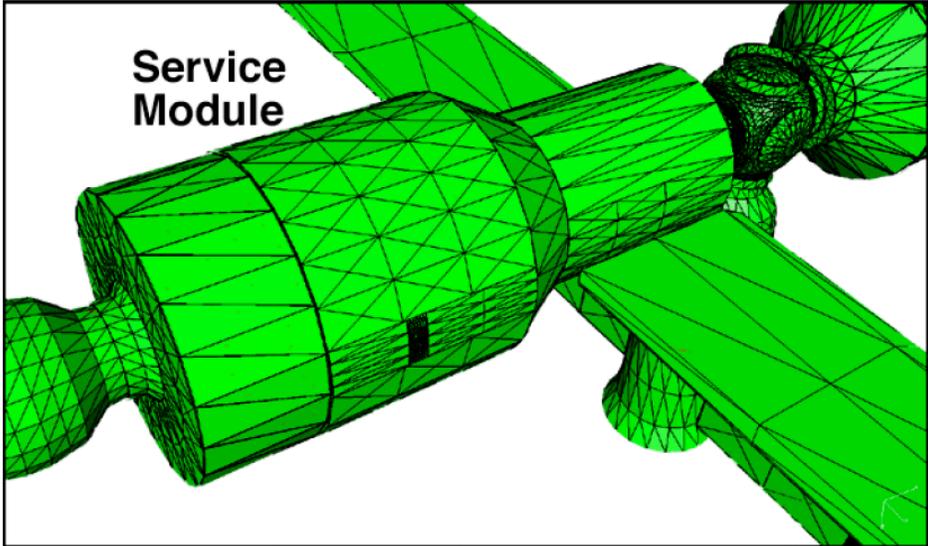
# ISS Ascending Node Anisotropic Proton Flux Distributions from AP8/AP9 Trapped Models at the Center of SAA



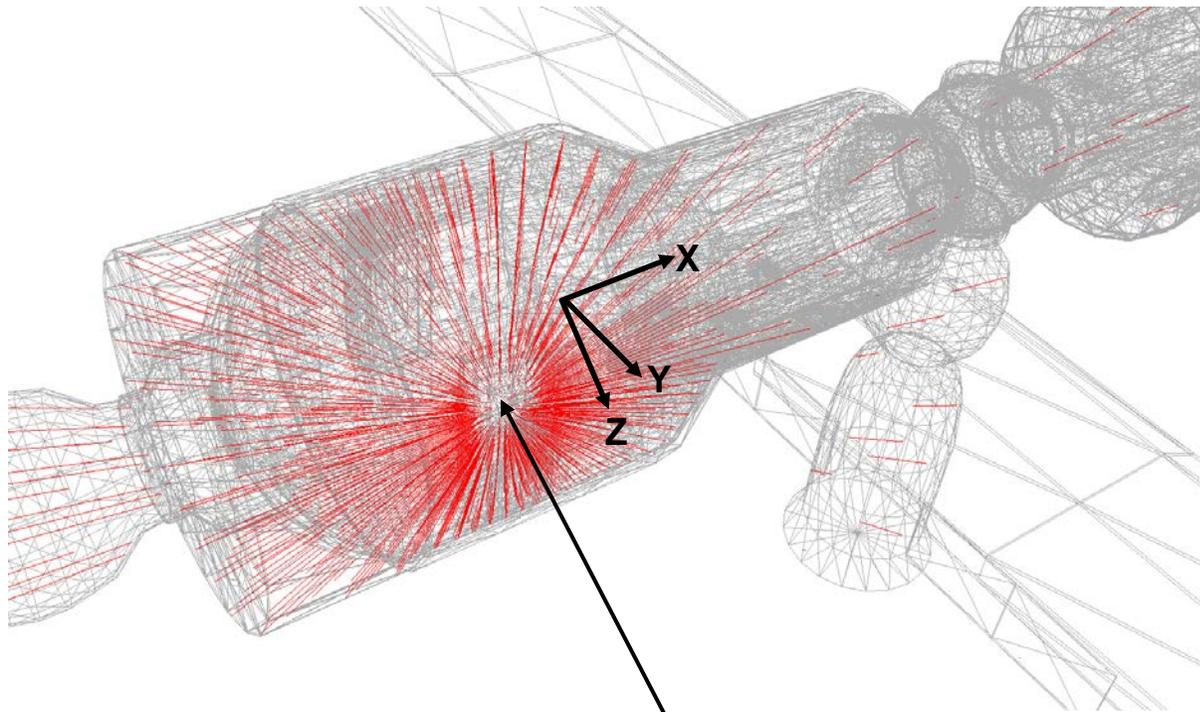
Here, azimuth and zenith are in ISS internal coordinate



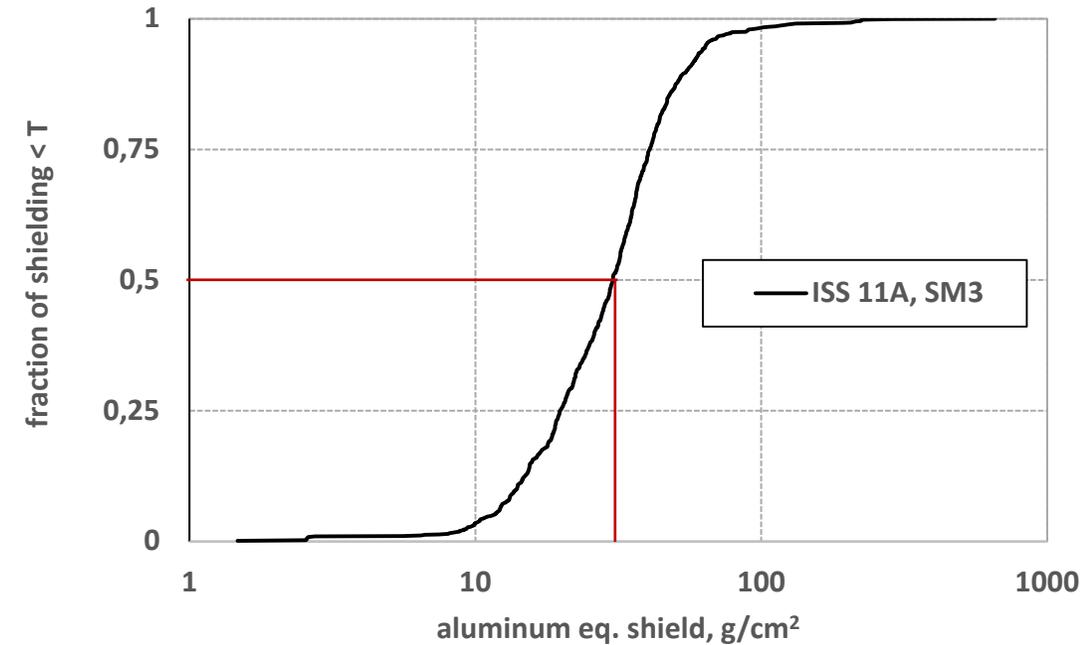
# ISS Dosemetric Target Point Definition



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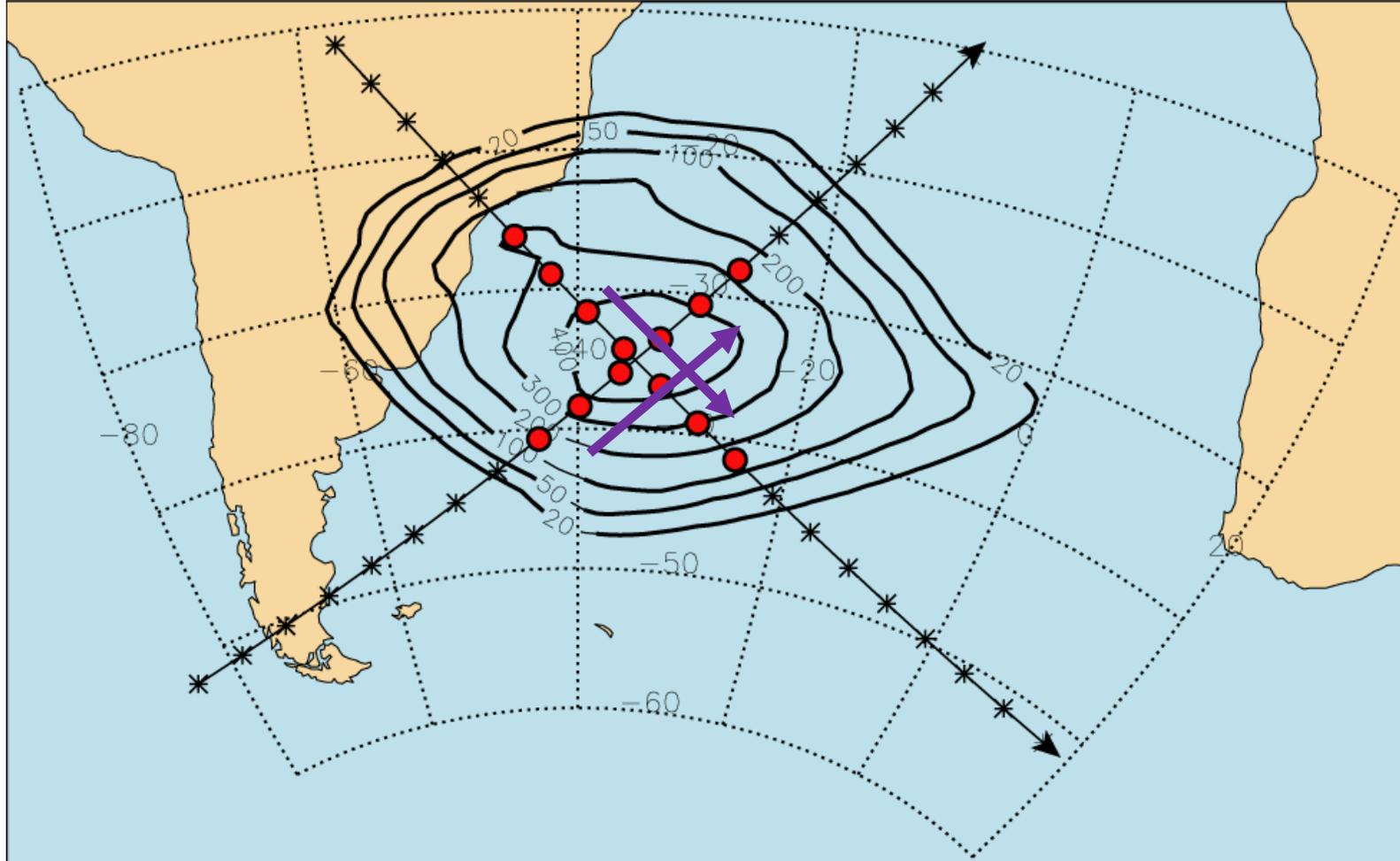


CQ ray traced at SM3 target pt.



SM3 target pt. thickness dist. of CQ (50% ~ 20 g/cm<sup>2</sup>)

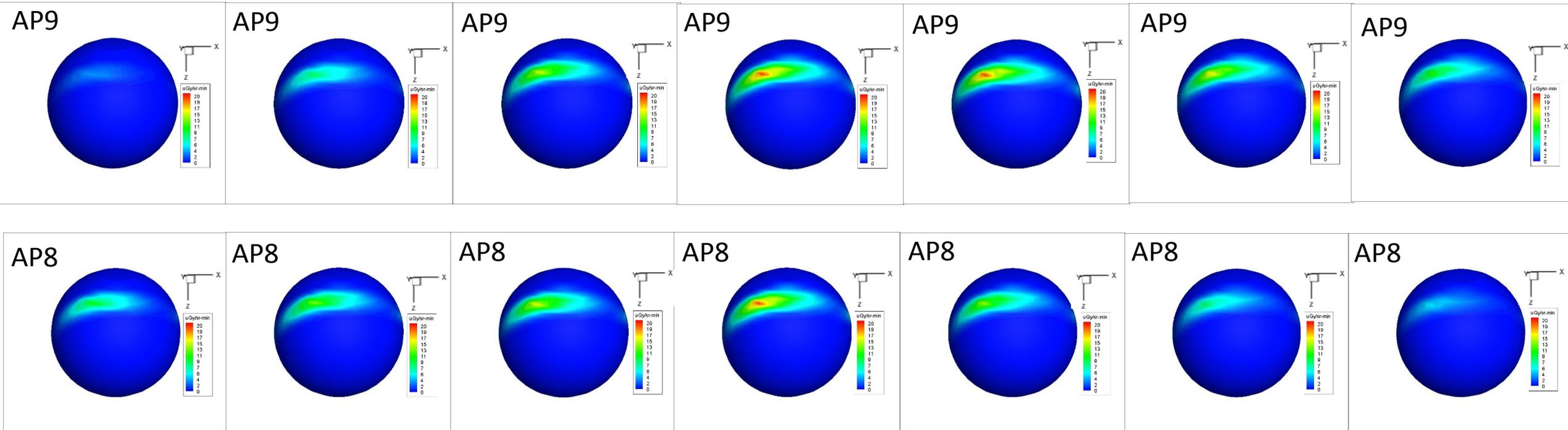
# ISS Ascending and Descending Nodes within SAA



# ISS Ascending Node AP8/AP9 Comparison within SAA



SM3 target pt. directional dose ( $\mu\text{Gy}/\text{sr}/\text{min}$ ) in a Silicon detector



min. 1

min. 2

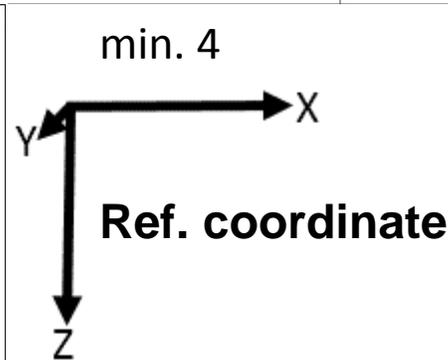
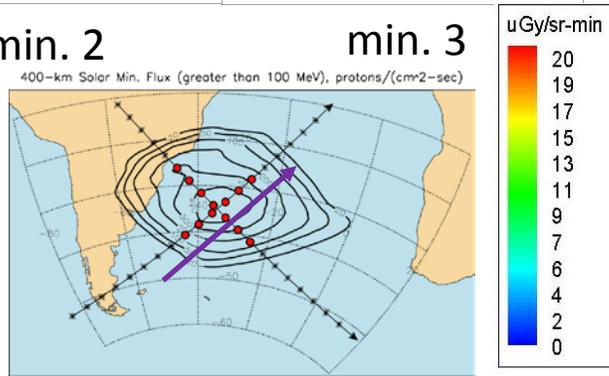
min. 3

min. 4

min. 5

min. 6

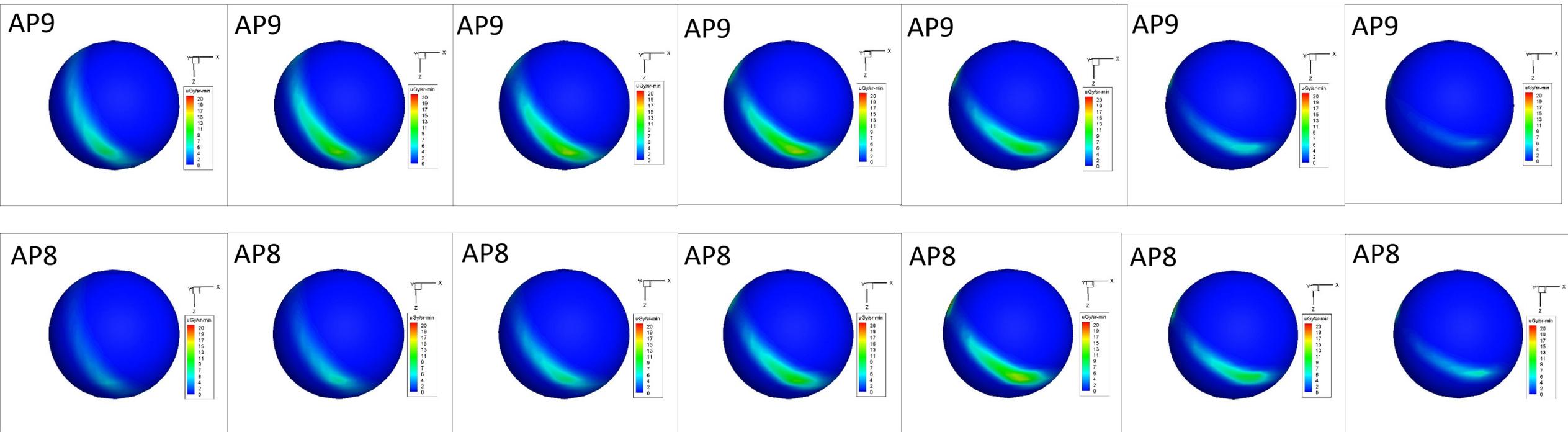
min. 7





# ISS Descending Node AP8/AP9 Comparison within SAA

SM3 target pt. directional dose ( $\mu\text{Gy}/\text{sr}/\text{min}$ ) in a Silicon detector



min. 1

min. 2

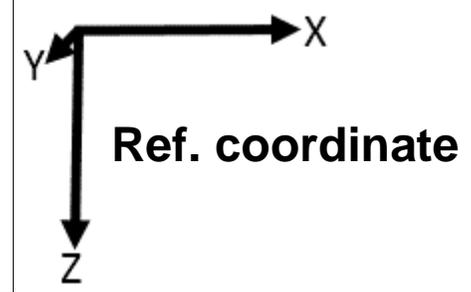
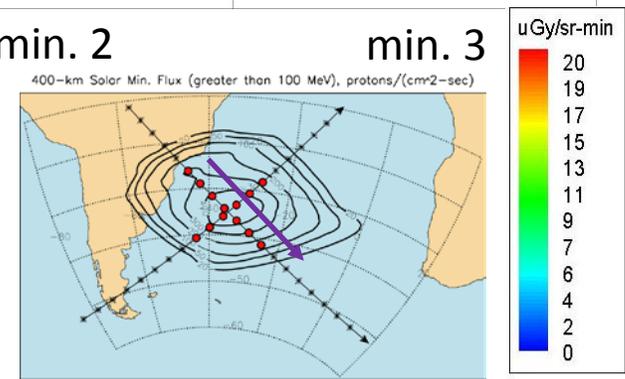
min. 3

min. 4

min. 5

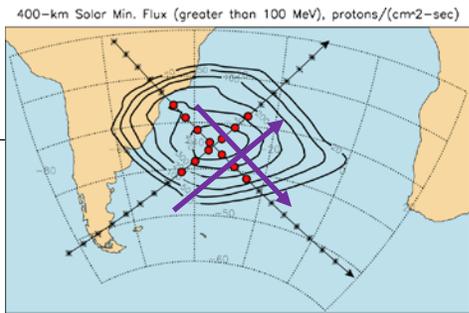
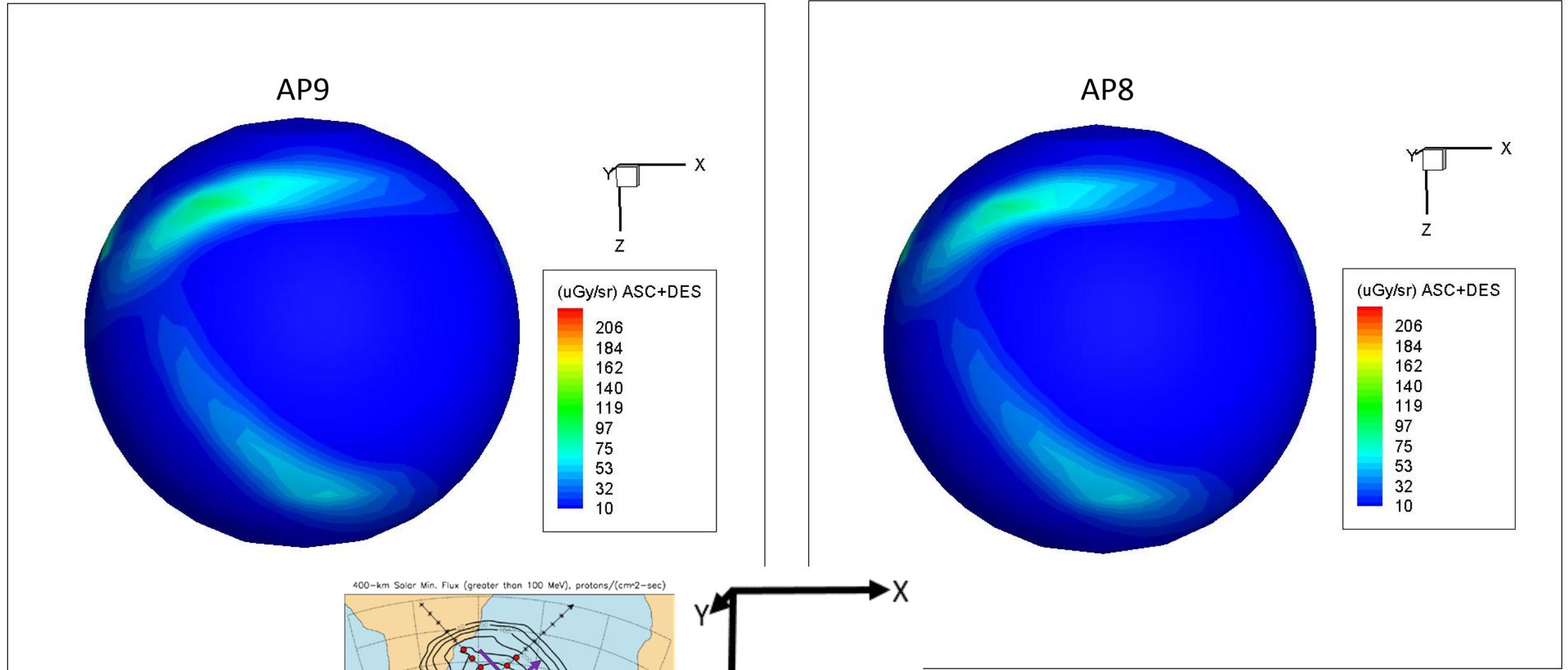
min. 6

min. 7



# ISS Combined Nodes AP8/AP9 Comparison within SAA

Ascending+descending nodes CQ-SM3 target pt. exposure for 7 mins ( $\mu\text{Gy}/\text{sr}$ )

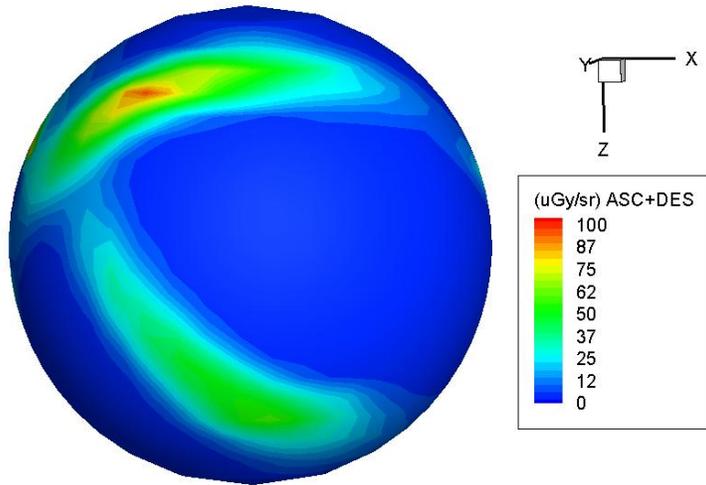


# ISS Combined Nodes AP8/AP9/GCR Comparison within SAA



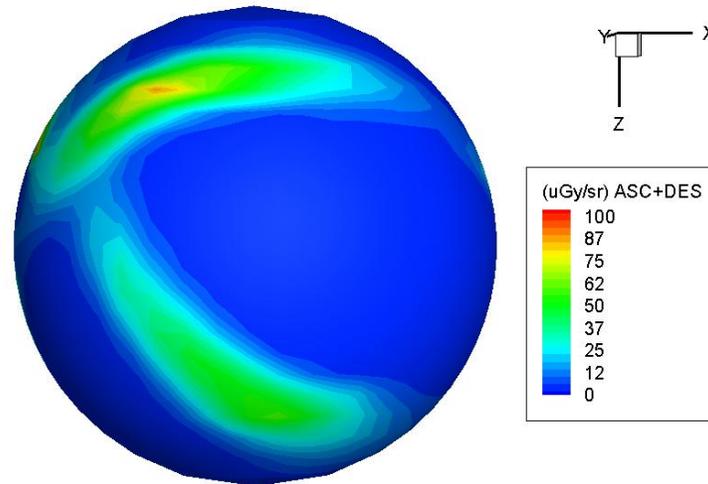
Ascending+descending nodes CQ-SM3 target pt. exposure for 7 mins ( $\mu\text{Gy}/\text{sr}$ )

AP9



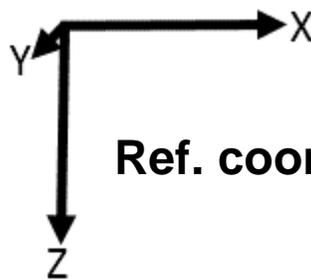
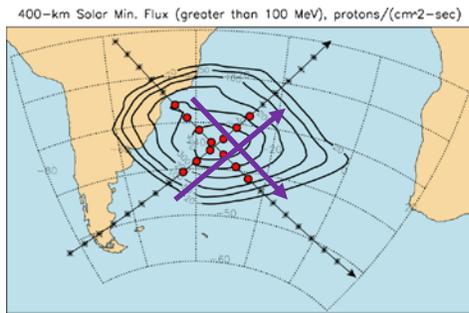
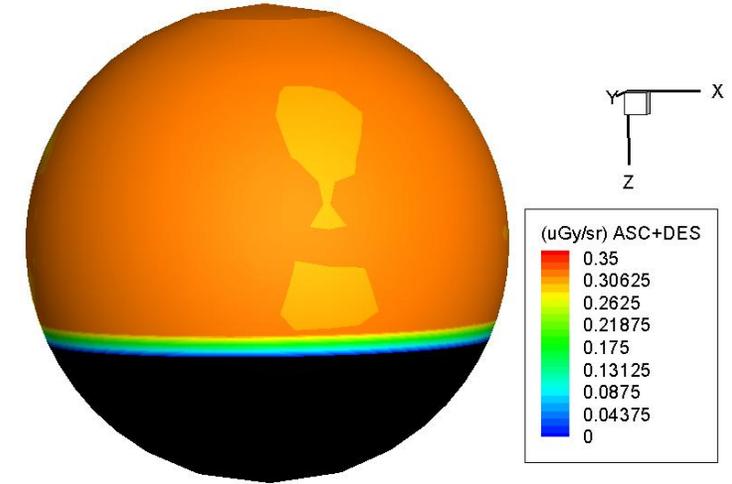
$[2.6\mu\text{Gy}/\text{sr}/\text{orbit} * 5 \text{ orbits}/\text{day} * 4\pi_{\text{sr}}]=161 \mu\text{Gy}/\text{day}$

AP8



$[2.2\mu\text{Gy}/\text{sr}/\text{orbit} * 5 \text{ orbits}/\text{day} * 4\pi_{\text{sr}}]=142 \mu\text{Gy}/\text{day}$

GCR



Simulation: SM3 pt. (2003) AP8=142  $\mu\text{Gy}/\text{d}$ , AP9=161  $\mu\text{Gy}/\text{d}$   
 Measurement: G. Reitz (DOSMAP-2005) : 74 ~ 215  $\mu\text{Gy}/\text{d}$



# Summary

- Described an angular model to account for proton flux anisotropy at LEO. The angular model has not been tested for the **difficult to analyze** inner electron belt yet
- Using the omni-directional trapped models AP8/AP9, validated the angular model for AF-TSX5 satellite proton measurements at three locations within SAA
- Using CQ of Zvezda-SM ray-traced geometry of ISS and the trapped models AP8/AP9, applied the angular model to proton anisotropy **dosimetric simulation** of ISS within SAA and compared with ISS daily measurement



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**Q/A**